

Errata

Unsteady Aerodynamics of Slender Delta Wings at Large Angles of Attack

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[J. Aircraft 12, 721-729 (1975)]

IN Eq. (5a), $b/2c_0$ should be substituted for b/c_0 .

In Eq. (6), 1.75 should be substituted for 1.72.

Figure 8 contains several errors. On the $C_{m\theta}$ scale the value 0.6 should be 0.8, and some of the data points and theoretical curves are misplotted. The correct figure is shown below.

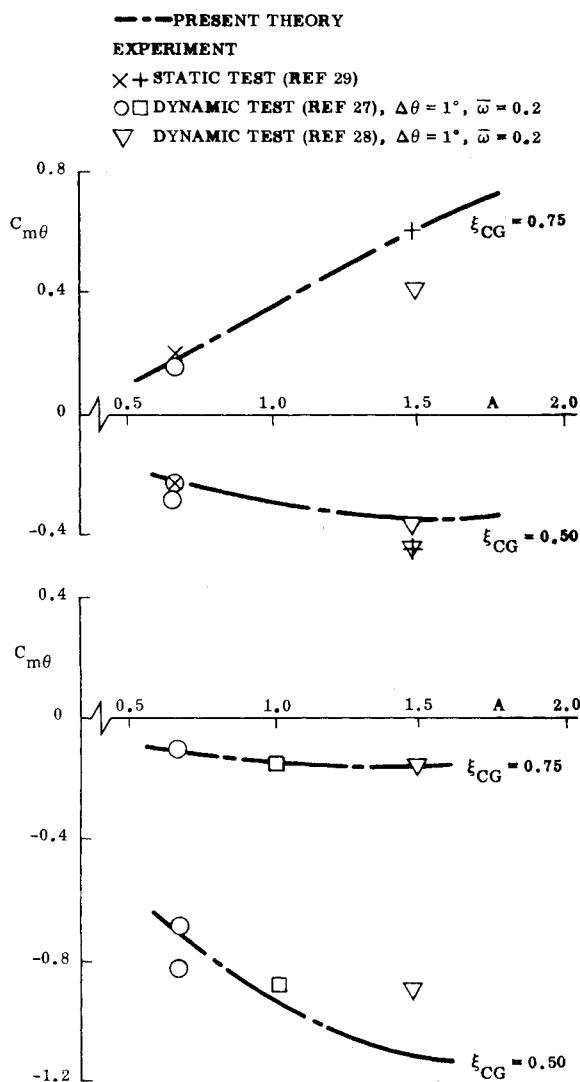


Fig. 8 Attached flow dynamic stability derivatives at $\alpha=0$ and $M=0$.

Received May 13, 1977.

Index categories: Aircraft Aerodynamics (including Component Aerodynamics); Aircraft Vibration.

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